# Apollo Mode A/C Transponder Model SL70 Installation Manual

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## **HISTORY OF REVISIONS**

Revision	Date	Description
Draft	8/11/99	Final draft.

## **IMPORTANT NOTE**

The conditions and tests required for TSO approval of this article are minimum performance standards. It is the responsibility of those desiring to install this article on or within a specific type or class of aircraft to determine that the aircraft operating conditions are within TSO standards. The article may be installed only if further evaluation by the applicant documents an acceptable installation and is approved by the Administrator.

## **ORDERING INFORMATION**

To receive additional copies of this publication, order part # **560-0402**, *Apollo SL70 Mode A/C Transponder Installation Manual*.

## REFERENCE PUBLICATIONS

Following are other publications referenced in this guide.

# TABLE OF CONTENTS

SECTION 1 - INTRODUCTION	1
ABOUT THIS MANUAL	
APOLLO SL70 DESCRIPTION	1
FEATURES	2
REGULATORY COMPLIANCE	2
UNPACKING THE EQUIPMENT	3
PACKAGE CONTENTS	3
OTHER REQUIRED MATERIALS	3
SPECIAL TOOLS REQUIRED	4
SECTION 2 - INSTALLATION	5
PRE-INSTALLATION INFORMATION	5
INSTALLATION OVERVIEW	5
INSTALLATION CONSIDERATIONS	5
MOUNTING CONSIDERATIONS	
MINIMUM SYSTEM CONFIGURATION	
ALTHUDE INPUT	
EQUIPMENT MOUNTING	
UNIT INSTALLATION / REMOVAL	
ELECTRICAL CONNECTIONS	
POWER	
ALTITUDE INPUTSERIAL INTERFACE	
DISCRETE INPUTS	
Antenna Installation and Connections	
POST INSTALLATION CHECKOUT	
TEST MODE CHECKOUT AND SETUP	12
OPERATION / PERFORMANCE CHECKOUT	13
SECTION 3 - SPECIFICATIONS	15
ELECTRICAL	15
Physical	15
Environmental	16
ALTITUDE INPUT	16
DISCRETE INPUTS	16
SERIAL INTERFACE	
Transponder Performance	
Antenna Requirements	17
SECTION 4 - LIMITATIONS	19
INSTALLATION	19
OPERATIONAL	19
APPENDIX A - TROUBLESHOOTING	
CONTACTING THE FACTORY FOR ASSISTANCE	22

APPENDIX B - PERIODIC MAINTENANCE	23
BIENNIAL CHECK	23
CLEANING THE FRONT PANEL	
APPENDIX C - ENVIRONMENTAL QUALIFICATIONS	25
APPENDIX E - SERIAL DATA PROTOCOL	27
ALTITUDE INPUT / OUTPUT	
LIST OF TABLES	
TABLE 1 PACKAGE CONTENTS	3
TABLE 2 REAR CONNECTOR PINOUT	
TABLE 3 TROUBLESHOOTING GUIDE	21
Table 4 Altitude Data Format	27
LIST OF FIGURES	
FIGURE 1 SL70 FRONT PANEL	2
FIGURE 2 MOUNTING FRAME ASSEMBLY	
FIGURE 3 CABLE ROUTING	7
FIGURE 4 COAX CABLE ASSEMBLY	9
FIGURE 5 WIRING DIAGRAM - GRAY CODE ALTITUDE INPUT	10
FIGURE 6 WIRING DIAGRAM - SERIAL ALTITUDE INPUT	
FIGURE 7 UNIT DIMENSIONS	

# **SECTION 1 - INTRODUCTION**

### ABOUT THIS MANUAL

This manual describes the installation of the Apollo SL70 transponder. It is intended for use by persons certified by the Federal Aviation Administration (FAA) to install aircraft avionics.

Section 1 Provides an **introduction** to the Apollo SL70 unit. TSO certification information is also included in this section.

Section 2 Includes **installation** and checkout procedures.

Section 3 Includes complete **specifications**.

Section 4 Includes **limitations** for the equipment and installation.

Appendix A Includes **troubleshooting** information.

Appendix B Includes **periodic maintenance** requirements.

Appendix C Includes the **environmental qualification form**.

Appendix E Includes **serial data protocol** specifications.

#### APOLLO SL70 DESCRIPTION

The Apollo SL70 is a TSO-C74c, Class A2, mode A/C transponder. The SL70, in addition to providing replies to ATC interrogations, includes an altitude display and altitude hold function and user-friendly interface features including rotary knob squawk code selection, a VFR button, and built-in self-test and diagnostics.

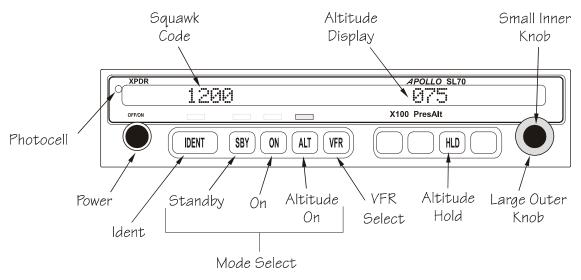


Figure 1 SL70 Front Panel

#### **FEATURES**

- small size 1.3" panel height
- full range input supply voltage
- high brightness LED display
- altitude display
- altitude hold function
- gray code or RS-232 serial data altitude input
- RS-232 altitude data output
- built-in self-test and diagnostics

## **REGULATORY COMPLIANCE**

The Apollo SL70 is designed and tested to meet the following TSO:

• FAA TSO-C74c, Class A2

The SL70 complies with the FCC requirements specified in:

• CFR 47, Part 87, Aviation Services, Subpart D, Technical Requirements

The SL70 software is designed and tested to RTCA/DO-178B, level C.

Note: Un-authorized changes or modifications to the SL70 may void the compliance to regulatory agency requirements and authorization for continued equipment usage.

## UNPACKING THE EQUIPMENT

Carefully unpack the equipment. Visually inspect the package contents for any evidence of shipping damage. Retain all shipping containers and packaging material in case reshipment is necessary.

#### PACKAGE CONTENTS

As shipped from the UPS Aviation Technologies factory, the Apollo SL70 package includes most necessary items for installation other than supplies normally available at the installation shop, such as wire and coax cable, the antenna, or any optional switches. The items included in the package are listed in Table 1.

Table 1 Package Contents			
Part #	Description	Qty	
Apollo SL70 Insta	allation Kit, Part # 424-0306-xx		
162-0103	37 pin dsub connector shell	1	
162-0043	Right angle coax plug	1	
202-0001	Cable tie	2	
204-0037	Edge grommet	6"	
221-0400	4-40 x <sup>1</sup> / <sub>4</sub> SS pan head Phillips machine screw with lock	10	
	washer		
224-0404	4-40 x 1/4 flat head Phillips machine screw	2	
240-0008	9/16 OD flat washer	1	
245-0022	Crimp contact for dsub, 20 to 24 awg wire	37	
265-0007	7/16" retaining ring	1	
310-2295-00	Connector mounting bracket	1	
310-5181-00	Mounting frame	1	
998-0048	3/32 hex driver	1	
Apollo SL70 Mar	nual Kit, Part # 564-0072-xx		
560-0401-00	SL70 User's Manual	1	
560-0402-00	SL70 Installation Manual	1	

# OTHER REQUIRED MATERIALS

In addition to the materials supplied with the SL70, the following is required.

- a suitable 2A circuit breaker
- an altitude data source either gray code or RS232
- a standard transponder antenna

# SPECIAL TOOLS REQUIRED

# **Crimp Tool**

A crimp tool meeting MIL specification M22520/1-01 and a positioner are required to ensure consistent, reliable crimp contact connections for the rear 37 pin dsub connector. These tools are available from:

Astro Tool Corp. Phone (503)642-9853 21615 SW TV Highway Fax (503)591-7766

Beaverton, OR 97006

Crimp tool: Astro Tool part #615708 Positioner: Astro Tool part #616356

# **SECTION 2 - INSTALLATION**

This section describes the installation of the Apollo SL70 including mounting, wiring, and connections. A post installation check-out procedure is included at the end of this section.

## **PRE-INSTALLATION INFORMATION**

Always follow good avionics installation practices per FAA Advisory Circulars AC 43.13-1B, 43.13-2A, or current FAA guidance.

Follow the installation procedure in this section as it is presented for a successful installation. Read the entire section before beginning the procedure. Perform the post installation check-out before closing the work area in case problems occur.

## **INSTALLATION OVERVIEW**

A successful installation should start with careful planning including determination of mounting location for the SL70, antenna location and mounting, connections to an altitude data source, power, cable routing, and other required modifications.

Once the mounting location has been determined, prepare the mounting frame for installation. If may be easier to complete the wiring harness and attach the connectors to the mounting frame prior to installing the mounting frame in the aircraft.

#### Installation Considerations

#### MOUNTING CONSIDERATIONS

The SL70 is designed to mount in the avionics stack in the aircraft instrument panel within easy view and reach of the pilot. The standard package includes the mounting frame for ease of mounting, connections, and service of the unit. Allow an additional one inch clearance to the rear of the mounting frame for connections and cables.

For typical installations, the SL70 does not require external cooling. When mounting the unit, leave a clearance of 1/8 to 1/4 inch between avionics to allow for air circulation.

#### MINIMUM SYSTEM CONFIGURATION

The SL70 requires connections to the following equipment as a minimum:

- power input
- altitude data gray code or RS232
- a standard transponder antenna

#### **ALTITUDE INPUT**

The SL70 can use either a standard altitude gray code input from an encoding altimeter or an RS-232 input from a serial encoder. For new installations, a serial encoder is recommended because it simplifies the wiring and is more reliable. The SL70 can detect failures on the RS-232 input, thereby eliminating erroneous mode C altitude replies.

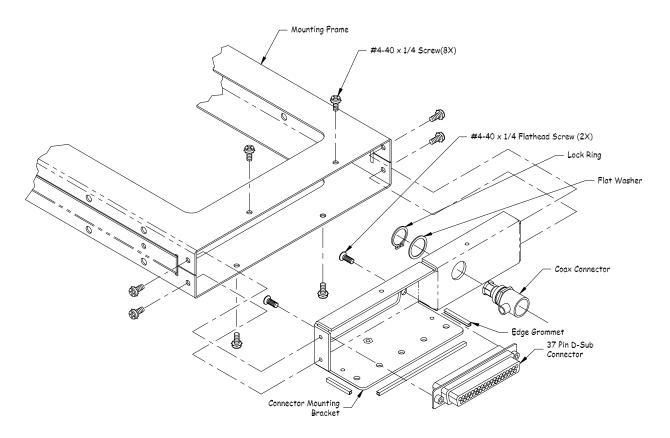
The altitude source must be accurate to within  $\pm 125$  feet, 95% probability, and have a resolution of 100 feet (the serial input resolution may be better than 100feet).

#### ANTENNA

The antenna shall be a standard transponder quarter wave monopole antenna, such as the TED #104-12 or Comant CI-101.

## **EQUIPMENT MOUNTING**

Once the cable assemblies have been made, attach the 37 pin dsub and coaxial cable connectors to the rear connector mounting plate and the mounting frame as illustrated in Figure 2 and Figure 3. Route the wiring bundle as appropriate. The rear connector plate should be attached to the mounting frame before installing the frame in the instrument panel. The rear connector plate can be used to tie down the cable assemblies. Connect the shields on any shielded cables directly to the connector mounting plate.



**Figure 2 Mounting Frame Assembly** 

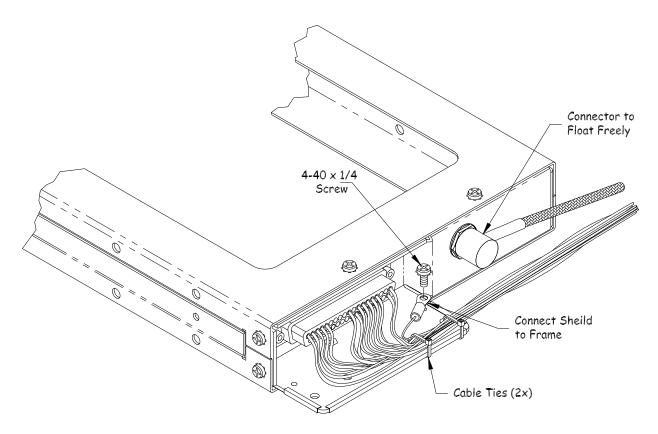


Figure 3 Cable Routing

Once the cable assemblies are complete and the connectors are attached to the mounting frame, install the mounting frame assembly in the aircraft instrument panel. Be sure to use low profile head screws so the unit will slide in and out freely. Attach the front of the mounting frame to the instrument panel. Use support brackets to attach the rear of the frame to the aircraft.

#### UNIT INSTALLATION / REMOVAL

To install the SL70 in the mounting frame, make sure the cam lock is rotated so the rear part is up, then slide the unit into the frame and tighten with the 3/32 hex tool. The unit will be pulled into the frame by the cam lock and the connectors will fully engage.

To remove the SL70 from the mounting frame, use the hex tool and turn the tool CCW. The unit will be pushed out of the frame by the cam lock assembly. No special extraction tools are required.

#### **ELECTRICAL CONNECTIONS**

The SL70 installation kit includes a 37 pin dsub shell and crimp contacts. The crimp contacts are specified for 20 to 24 AWG wire. Make the crimp connections with a crimp tool as specified in the Special Tools Required section on page 4. All wires should be 20 to 24 AWG unless otherwise specified. Wiring diagrams are included on pages 10 and 11.

#### Power

The SL70 is internally fused at 3 amps. A separate 2 amp circuit breaker or fuse should be installed for downline overload or short circuit protection. Make the power connections to the SL70 using 20 AWG wire.

**Note:** Circuits should be protected in accordance with guidelines in AC 43.13-1B, chapter 11, section 4.

#### **ALTITUDE INPUT**

If using the gray code altitude input, connections can be made directly from the SL70 to the altitude source. Isolation diodes are included within the SL70. Make sure a common ground connection exists, or connect a ground wire between the source and the SL70.

If using the RS-232 serial altitude input, make the connections as described in the following section.

#### **SERIAL INTERFACE**

The SL70 includes an RS-232 serial port which can be used for inputting or outputting altitude data. This is an optional connection if the altitude gray code input is used.

When making serial connections to the SL70, use a shielded two or three conductor cable. Make the RxD, TxD, and serial ground connections on the 37 pin dsub connector. Connect the shield to the rear of the mounting frame on the connector plate.

Complete serial interface specifications are included in Appendix E.

#### **DISCRETE INPUTS**

#### **Remote Ident**

The ident input can be connected to a remotely mounted momentary switch to initiate the ident function the same as using the front panel IDENT button.

When using the ident input, connect it to a remotely mounted momentary switch, and connect the other terminal of the switch to ground.

#### **Remote Standby**

The standby input is for use when the installation includes more than one transponder. When two transponders are installed, connect the standby input to a two position selector switch, with the common on the switch connected to ground. The SL70 will be in standby when the input is pulled low to ground.

#### **Suppresion Input**

The suppress input is used when installed with a DME that includes a suppression output. The transponder will be suppressed, or will not generate replies, when the input is driven high from the DME (or other source). To use the suppression input, connect it to the suppression output on a DME (or other appropriate suppression source).

### ANTENNA INSTALLATION AND CONNECTIONS

The antenna should be mounted in a vertical position in an area on the bottom of the aircraft away from other antennas or landing gear.

The antenna coax cable should be a double shielded low loss cable and must have a cable loss of < 3 dB, including connectors. A typical installation should have a cable loss of 1½ to 2 dB. Several suitable coaxes are RG142B and RG400.

The assembly of the rear panel coax connector included with the installation kit is illustrated in the following figure.

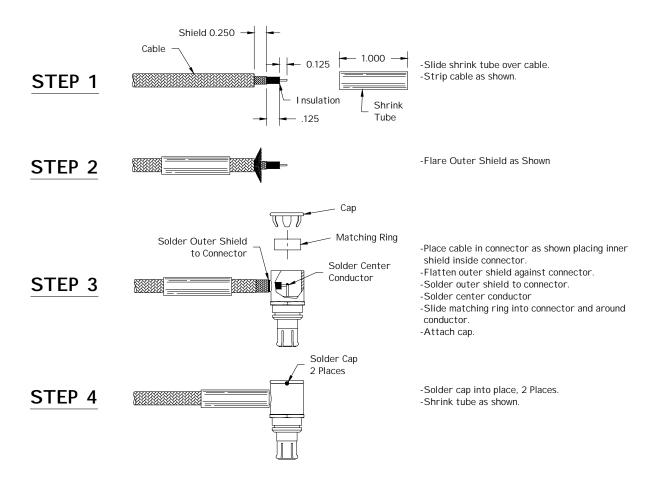
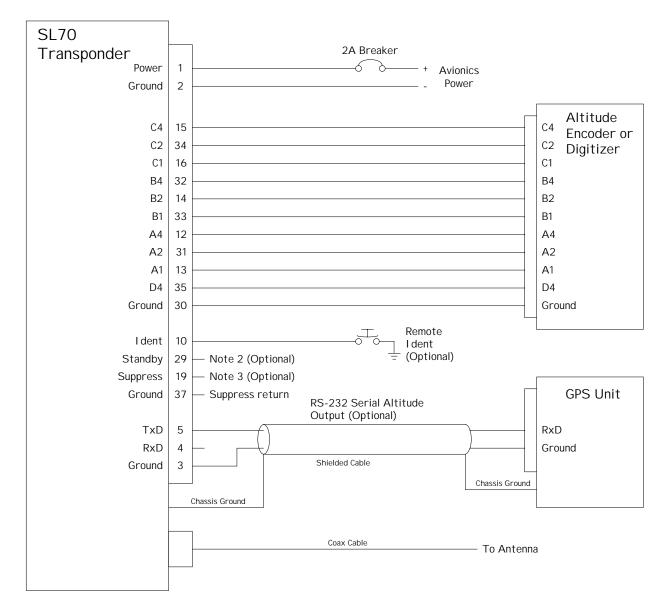


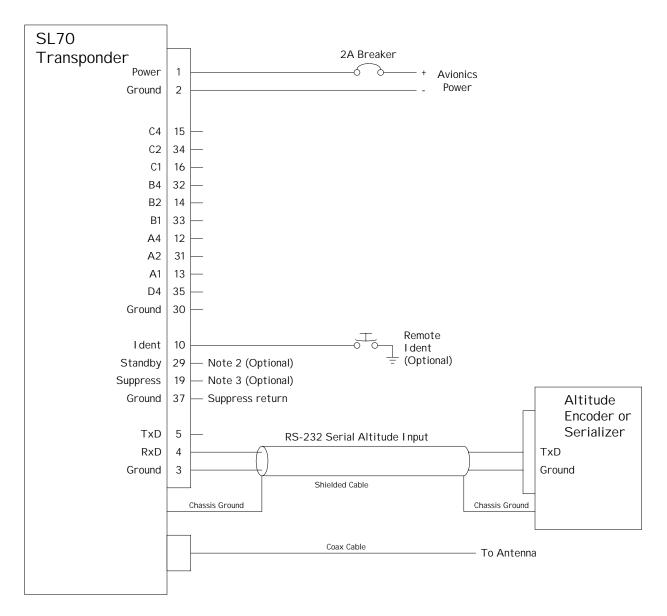
Figure 4 Coax Cable Assembly



#### Notes:

- 1. Select the gray code input in the setup mode when using the parallel gray code altitude input.
- 2. Use standby input to select between two transponders using a selector switch. Unit is in standby when input switched to ground.
- 3. Suppression input, typically connected to DME suppression output. Transponder is suppressed when input is high.

Figure 5 Wiring Diagram - Gray Code Altitude Input



#### Notes:

- 1. Select altitude input in the setup mode when using RS-232 serial altitude input.
- 2. Use standby input to select between two transponders using a selector switch. Unit is in standby when input switched to ground.
- 3. Suppression input, typically connected to DME suppression output. Transponder is suppressed when input is high.

Figure 6 Wiring Diagram - Serial Altitude Input

#### POST INSTALLATION CHECKOUT

Once the unit is installed, complete the checkout procedure to verify proper operation. Refer to the user's guide for operating instructions.

The steps that are not applicable to a particular installation may be skipped. A checkout log sheet is included on page 14 to fill out during the checkout procedure. Make a photocopy of the log sheet for ease of use if desired.

#### **Mounting / Wiring Check**

Verify that all cables are properly secured and shields are connected to the rear of the mounting frame. Check the movement of the aircraft controls to verify that there is not interference.

#### TEST MODE CHECKOUT AND SETUP

The SL70 has a built-in setup mode to simplify the checkout. To operate the SL70 in the setup mode, hold down the "IDENT" and "ALT" buttons while switching on the power. To return to normal operation, switch the power off, then back on.

## **Altitude Input Source Selection**

The SL70 altitude input type must be selected. To select the altitude input type:

- 1. In test mode, rotate the LARGE knob to the "SL70 CONF" (SL70 Configuration) page, then rotate the SMALL knob to the "ASRC" (Altitude source) page.
- 2. Press IDENT (the altitude type field will start to flash), rotate the SMALL knob to select the altitude type, then press IDENT to save the selection.

The altitude types that can be selected are:

GRAY	to use the parallel gray code input	
SER	to use the RS-232 serial altitude input	

#### **RS-232 Baud Rate Selection**

If using the RS-232 serial interface, the baud rate must be selected to match the connected equipment. To select the baud rate:

- 1. In test mode, rotate the LARGE knob to the "SL70 CONF" (SL70 Configuration) page, then rotate the SMALL knob to the "BAUD" (baud rate) page.
- 2. Press IDENT (the baud rate field will start to flash), rotate the SMALL knob to select the desired baud rate, then press IDENT to save the selection.

The baud rates available are 1200, 2400, 4800, 9600, and 19200.

#### **VFR Code Selection**

The VFR code used by the SL70 when pressing the "VFR" button can be selected as appropriate for the aircraft's operating area. To change the VFR code:

- 1. In test mode, rotate the LARGE knob to the "SL70 CONF" (SL70 Configuration) page, then rotate the SMALL knob to the "VFR" (VFR code selection) page.
- 2. Press IDENT (the first VFR code character will start to flash).

- 3. Rotate the SMALL knob to change the character, rotate the LARGE knob to move the cursor.
- 4. Press IDENT to save the selection.

#### **OPERATION / PERFORMANCE CHECKOUT**

#### Self Test

The SL70 includes a self test that is executed every time the unit is turned on that checks the receiver and transmitter operation as well as other internal functions.

Verify that the unit does not display a failure indication when turned on.

### **Altitude Input**

Verify that the displayed altitude matches the altimeter pressure altitude (at 29.92).

#### **External Inputs**

If the external ident or standby inputs are connected, verify operation by:

- a) Verify that the unit goes to standby when the external standby input is pulled low.
- b) Verify that the ident LED turns on when the external ident button is pressed (must be in the "ON" or "ALT" modes).

#### Performance (Ramp) Test

After installation, the transponder should be tested as specified in Appendix F of CFR 14 part 43, to AC 43-6A, and/or other appropriate regulations. The test is typically done as a ramp test using a transponder ramp test set, such as the IFR ATC-600A. The ramp test includes checks as follows.

Reference part 43 Appendix F:

- a) reply frequency
- b) suppression
- c) receiver sensitivity
- d) reply RF output power

Reference AC 43-6A:

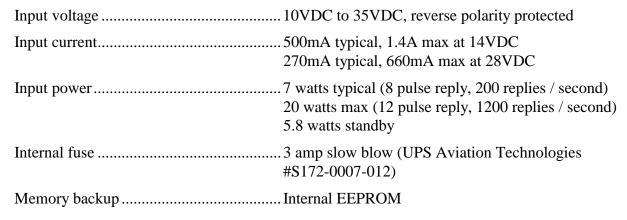
altitude reporting

APOLLO SL70 POST-INSTALLATION C	HECKOUT LOG Date:// By:
CONFIGURATION INFORMATION:	
Apollo SL70 430-6090 Mod	
Serial #	
TEST MODE CHECKOUT AND SETUP:	
Self Test:	
[] Pass	
[] Fail	DG 444 1
Altitude source:	RS-232 baud rate:
[] Gray code (GRAY)	VFR Code:
[] Serial RS-232 (SER)	
OPERATION / PERFORMANCE CHECK:	
Altitude data (on display):	Performance verification:
[] Pass	[] Pass
External inputs:	[] 1 455
[] Remote ident checked [] N/A	
[] External standby [] N/A	
[] External standoy [] IVA	
COMMENTS:	

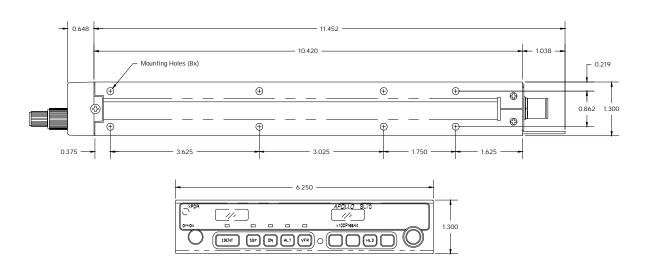
# **SECTION 3 - SPECIFICATIONS**

This section includes detailed electrical, physical, environmental, and performance specifications for the Apollo SL70.

## **ELECTRICAL**



## **PHYSICAL**



**Figure 7 Unit Dimensions** 

#### **ENVIRONMENTAL**

The Apollo SL70 is designed and tested to meet appropriate categories of RTCA/DO-160D. The Environmental Qualification Form is included in Appendix C.

Operating temperature.....-20°C to +55°C

Storage temperature.....-55°C to +85°C

Cooling...... Not required

## **ALTITUDE INPUT**

isolation diodes.

Range: -1000feet to 63,000 feet On: <= 3.5 volts, Off: open

Serial Interface Specifications)

**Note:** The altitude data input type must be selected using the setup function during the post installation checkout.

#### **DISCRETE INPUTS**

Remote Ident ...... Input pulled low momentarily to initiate ident

transmission (same function as front panel button)

On: <= 3.5 volts, Off: open

Standby input...... Input pulled low to disable the transponder (will not

generate replies)

On: <= 3.5 volts, Off: open

Suppression input ...... Input pulled high to initiate suppression – typically

connected to DME suppression output

Suppressed: >= 5.0 volts;

Not suppressed: < 2.5 volts (or open)

#### SERIAL INTERFACE

RS-232..... Defined in Appendix E – Serial Interface

**Specifications** 

# TRANSPONDER PERFORMANCE

TSO	.TSO-C74c
TSO Class	.2A
Warm-up	none required
Receiver Frequency	.1030 MHz
Sensitivity (MTL)	.–72 dBm
Dynamic Range	.> 50 dB
Side Lobe Suppression	.2 pulse (P1,P2)
Transmitter Frequency	.1090 MHz +/- 120 kHz
Transmitter Power	.250 watts minimum, 325 watts typical
Reply rate	.1200 / second
Mode A Capability	.4096 codes plus SPI ident pulse
Mode C Capability	-1000 to 63,000 feet, 100 foot increments, from either gray code or RS-232 serial inputs

## **ANTENNA REQUIREMENTS**

The SL70 requires a standard transponder quarter wave monopole antenna, such as the TED #104-12 or Comant CI-101.

		Ta	able 2 Rear Connector Pinout
Pin#	I/O	Connection	Function
1	I	Power +	main DC power input
2	I	Power ground	main DC power ground
3	О	Serial ground	RS-232 serial ground
4	I	RxD	RS-232 serial data input
5	О	TxD	RS-232 serial data output
6	О	reserved	•
7	О	reserved	
8	I	reserved	
9	I	reserved	
10	I	Ident	remote ident input
11		NC	no connection
12	I	A4	A4 altitude gray code input
13	I	A1	A1 altitude gray code input
14	I	B2	B2 altitude gray code input
15	I	C4	C4 altitude gray code input
16	I	C1	C1 altitude gray code input
17		reserved	
18		reserved	
19	I	Suppress	remote suppression input
20	О	NC	no connection
21	I	reserved	
22	O	reserved	
23		NC	no connection
24		Ground	
25	I	reserved	
26	I	reserved	
27		Ground	
28		Ground	
29	I	Standby	remote standby input
30		Ground	
31	I	A2	A2 altitude gray code input
32	I	B4	B4 altitude gray code input
33	I	B1	B1 altitude gray code input
34	I	C2	C2 altitude gray code input
35	I	D4	D4 altitude gray code input
36		reserved	
37		Ground	



Viewed from rear of unit

# **SECTION 4 - LIMITATIONS**

## **INSTALLATION**

Aircraft installation must be made in accordance with this installation manual and applicable FAA FARs and ACs.

#### **OPERATIONAL**

The SL70 must be operated within the limitations as follows, or by other regulatory guidance as appropriate.

- 1. The altitude should be verified by comparing the pressure altitude display to the aircraft altimeter relative to barometric setting of 29.92.
- 2. The transponder is to be operated in compliance with CFR 14 section 91.215 for ATC transponder and altitude reporting equipment.
- 3. The transponder must be tested as specified in CFR 14 section 91.413 within the previous 24 months.

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# Notes

# **APPENDIX A - TROUBLESHOOTING**

This appendix provides information to assist troubleshooting if problems occur after completing the installation. Use Table 3 to assist in troubleshooting.

	Table 3 Troubleshooting Gu	ide
Problem	Cause	Solution
The SL70 does not power on.	The unit is not getting power.	Check power connections, breakers, and main avionics switch.
The altitude is not correct.	The unit is not getting the correct altitude.	If the gray code inputs are used, check all connections from the altitude source and/or verify the altitude source.  If the serial altitude input is used, check for the correct baud rate and connections to the altitude source.
The altitude display is ""	The unit is not getting an altitude, or the altitude is invalid.	Check the connections to the altitude source. Check the altitude source to make sure it is operating correctly.
The unit fails the start-up test.	TX fails.	Check antenna connections.
	Other failure.	The unit may need repair.
The unit does not generate replies.	The unit is in standby.	The unit must be in either the "On" or "Alt" modes to generate replies.
	The unit is not receiving interrogation signals.	Check the antenna connections. If the unit does not "Fail" the receive test, then check antenna connections.
	No interrogation signal available.	Need to either use a ramp test set or climb to altitude where ATC radar service is available.

## CONTACTING THE FACTORY FOR ASSISTANCE

If the Apollo SL70 unit fails to operate despite troubleshooting efforts, contact the UPS Aviation Technologies factory for assistance.

UPS Aviation Technologies 2345 Turner Rd. S.E. Salem, Oregon 97302 U.S.A.

Phone (503)581-8101 or 1-800-525-6726

Be prepared to offer the following information about the installation:

- Installation configuration (accessories, antenna, ...)
- Model number, part number with mod levels, and serial number
- Software versions (The software versions can be displayed in the setup mode by rotating the LARGE knob to the "SW VER" page, and rotating the SMALL knob to display the microcontroller and FPGA versions.)
- Description of problem
- Efforts made to isolate the problem

# APPENDIX B - PERIODIC MAINTENANCE

The Apollo SL70 is designed to not require any regular general maintenance except as included in this section.

## **BIENNIAL CHECK**

The transponder must be tested within the previous 24 months as specified in 91.413 of the FAA regulations.

No other periodic maintenance is required.

## **CLEANING THE FRONT PANEL**

The front bezel, keypad, and display can be cleaned with a soft cotton cloth dampened with clean water. DO NOT use any chemical cleaning agents. Extreme care must be taken to avoid scratching the surface of the display.

# Notes

# APPENDIX C - ENVIRONMENTAL QUALIFICATIONS

The Apollo SL70 has been tested to the following environmental categories per procedures defined in RTCA/DO-160D.

E	nvironmen	tal Qualification Form
Nomenclature: Apollo SL70		Manufacturer:
Part No.: 430-6090-xxx		UPS Aviation Technologies
TSO No.: TSO_C74c		2345 Turner Road S.E.
		Salem, Oregon 97302
Conditions	Section	<b>Description of Conducted Tests</b>
Temperature and Altitude	4.0	Equipment tested to Category B1 with
In-flight Loss of Cooling	4.5.4	No cooling required
Altitude	4.6.1	Equipment tested to 25,000 feet
Decompression	4.6.2	Equipment tested for decompression to 35,000 feet
Overpressure	4.6.3	Equipment tested for overpressure
Temperature Variation	5.0	Equipment tested to Category C, 2°C/min
Humidity	6.0	Equipment tested to Category A, standard humidity
		environment
Operational Shocks and Crash	7	Equipment tested for both operational and crash safety
Safety		shocks. (Equipment operates normally after both the crash
•		safety shocks.)
Vibration	8.0	Equipment tested without shock mounts to Categories S(M)
Explosion Proofness	9.0	Equipment identified as Category X, no test required
Waterproofness	10.0	Equipment identified as Category X, no test required
Fluids Susceptibility	11.0	Equipment identified as Category X, no test required
Sand and Dust	12.0	Equipment identified as Category X, no test required
Fungus Resistance	13.0	Equipment identified as Category X, no test required
Salt Spray	14.0	Equipment identified as Category X, no test required
Magnetic Effect	15.0	Equipment is Class Z
Power Input	16.0	Equipment tested to Categories A (28volt systems) & B (14
•		and 28 volt systems)
Voltage Spike	17.0	Equipment tested to Category A
Audio Frequency Conducted	18.0	Equipment tested to Categories A & B
Susceptibility – Power Inputs		
Induced Signal Susceptibility	19.0	Equipment tested to Category Z
Radio Frequency Susceptibility	20	Equipment tested to Category V
(Radiated and Conducted)		
Emission of Radio Frequency	21	Equipment tested to Category M
Energy		
Lightning Induced Transient	22.0	Equipment tested to A3 and B2
Susceptibility		
Lightning Direct Effects	23.0	Equipment identified as Category X, no test required
Icing	24.0	Equipment identified as Category X, no test required
Electrostatic Discharge (ESD)	25.0	Equipment tested to Category A
Remarks:	1	

Environmental Qua	lifications
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# Notes

# APPENDIX E - SERIAL DATA PROTOCOL

This appendix includes the RS-232 serial port interface specifications.

#### **ALTITUDE INPUT / OUTPUT**

The SL70 will output altitude data whenever it is available, either from the gray code or serial input.

The SL70 will accept altitude input data when the serial altitude data source is selected.

The format for the altitude input and output data is as follows.

Baud rate
Data bits8
Stop bits1
Paritynone
Message length17 characters
Expected input rateonce per second
Output rateonce per second

Table 4 Altitude Data Format			
Byte	<b>Data Format</b>	Description	
1	" <b>#</b> "	ASCII "#" (023h)	
2	"A"	ASCII "A" (041h)	
3	"L"	ASCII "L" (04Ch)	
4	"	ASCII space (020h)	
5	"+" or "-"	Altitude sign: ASCII "+" or "-" (02Bh or 02Dh)	
6-10	ddddd	Altitude in feet, right justified, with leading zeros	
11	"T"	ASCII "T" (054h)	
12	"+" or "-"	Temperature sign: ASCII "+" or "-" (02Bh or 02Dh)	
13-14	dd	Altimeter temperature	
15-16	dd	Checksum of bytes 1 through 14, computed in hex, output is	
		ASCII format (i.e., "FA" hex)	
17	<cr></cr>	ASCII carriage return (0Dh)	

The SL70 will interpret several altitude input codes as status or error codes, in the place of the altitude data, as follows.

"-09980"...... Encoder heater not ready, expected during encoder warmup or if there is a loss of signal from the encoder. The SL70 will reply with only framing pulses to mode C interrogations.

"-09981"	. Possible hardware problem, expected from the encoder indicating a temperature above normal. The SL70 will reply with only framing pulses to mode C interrogations.
"-09982"	. Altitude out of range, expected from the encoder indicating that the altitude is out of a valid range. The SL70 will reply with only framing pulses to mode C interrogations.

When the gray code input is selected, the SL70 will output the "-09982" error code in place of the altitude data if an undefined altitude input is detected on the gray code input.

# Notes

